

Notes:

PHYSICS 12

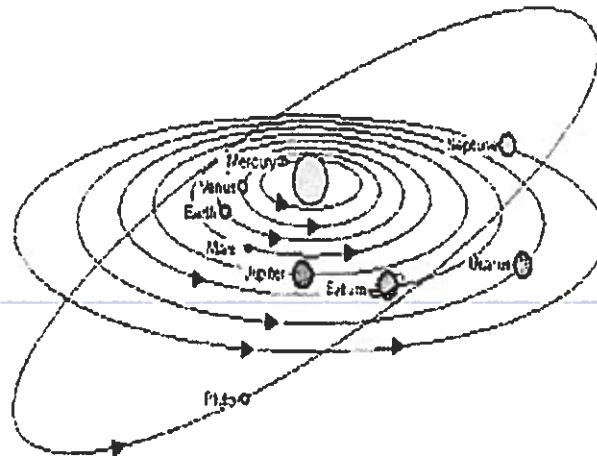
ORBITS

The object is specifically stated to be in orbit or it is located in space.

To find force, Newton's Law of Universal Gravitation must be used:

$$F_g = G \frac{m_1 m_2}{r^2}$$

$$G = 6.67 \times 10^{-11} \text{ N}\cdot\text{m}^2\text{kg}^{-2}$$



$$F_g = F_c$$

- Be careful not to substitute object's altitude = surface-to-surface distance but always center-to-center

Example: Find the speed of Apollo 13 when it is in orbit about the Moon at an altitude of 3000km.

$$m_M = 7.35 \times 10^{22} \text{ kg}$$

$$r_M = 1.74 \times 10^6 \text{ m}$$

$$\text{altitude} = 3000 \text{ km} \rightarrow 3.0 \times 10^6 \text{ m}$$

$$\text{orbiting radius} = r_M + \text{altitude} \quad (\Rightarrow \text{neglecting the size of Apollo 13})$$

$$F_c = F_g$$

$$\frac{v^2}{r} = \frac{G m_M m_A}{r^2}$$

$$V = \sqrt{\frac{(6.67 \times 10^{-11})(7.35 \times 10^{22})}{[1.74 \times 10^6 + 3.0 \times 10^6]}}$$

$$V = 1016.99 \dots \text{ m/s}$$

$$V^2 = \frac{G m_M}{r}$$

$$V = \sqrt{\frac{G m_M}{r}}$$

\therefore Apollo's speed is $1.02 \times 10^3 \text{ m/s}$.

1. a) Find the altitude of a satellite orbiting the Earth at 5000 m/s. Mass of the Earth is 5.98×10^{24} kg.

$$N = 5000 \text{ m/s}$$

$$m_E = 5.98 \times 10^{24} \text{ kg}$$

$$r_E = 6.38 \times 10^6 \text{ m}$$

$$\text{altitude} = r - r_E$$

$$N = \sqrt{\frac{G m_E}{r}}$$

$$N^2 = \frac{6 m_E}{r}$$

$$r = \frac{6 m_E}{N^2}$$

$$r = \frac{(6.67 \times 10^{-11})(5.98 \times 10^{24})}{(5.0 \times 10^3)^2}$$

$$r = 15954640 \text{ m}$$

$$\begin{aligned} \text{altitude} &= 15954640 - 6.38 \times 10^6 \\ &= 9.6 \times 10^6 \text{ m} \end{aligned}$$

\therefore Satellite's altitude is $9.6 \times 10^6 \text{ m}$.

b) Find the period of the probe.

Period = T = time to complete 1 revolution

$$t = \frac{d}{N}$$

$$= \frac{2\pi r}{N}$$

$$= \frac{2\pi (15954640)}{5000}$$

$$= 200495$$

$$6 = 5.6 \text{ hrs}$$

\therefore The probe completes one revolution around the Earth in 5.6 hours (or 2.0×10^4 s).

POTENTIAL ENERGY ASSOCIATED WITH GRAVITY

(for objects high above the Earth's surface)

GPE = gravitational potential energy

- This energy is found by evaluating work required to move an object from Earth's surface to infinity. This work is done by applied force that counters gravity.

$$GPE = -\frac{G m M}{r}$$

- GPE = 0 in infinity as objects separated by infinitely large radius do not interact.

Recall: Total energy of an object is given by the sum of the potential and kinetic energy of the object.

- For an object to escape Earth, the total energy has to be zero.
- Total energy of an orbiting satellite is always negative as the satellite is not free to move away.

$$E_{TOT} = KE + GPE$$

Derive the formula for escape velocity: $\Rightarrow E_{TOT} = 0 \text{ J}$

$$\begin{aligned} KE + GPE &= 0 \\ KE &= GPE \\ \frac{1}{2}mv^2 &= \frac{Gm_E m_2}{r} \end{aligned} \quad \left. \begin{aligned} v^2 &= \frac{2Gm_E}{r} \\ v &= \sqrt{\frac{2Gm_E}{r}} \end{aligned} \right\}$$

Escape Velocity

$$v = \sqrt{\frac{2Gm_E}{r}}$$

2. Read textbook p. 144-145 Chapter 5.5

3. a) What is the speed a satellite needs to maintain an orbit with a fixed radius?

$$\begin{aligned} F_g &= F_c \\ \frac{G m_E m_2}{r^2} &= m_2 \frac{v^2}{r} \\ G m_E &= v^2 \end{aligned}$$
$$v = \sqrt{\frac{G m_E}{r}}$$

b) How does the magnitude of this speed depend on the mass of the satellite?

- N_{orb} is mass independent.
- N_{orb} is dependent on the mass of the object that creates the gravitational field and the orbiting radius.

4. What is a geosynchronous satellite?

A satellite whose period is the same time needed for Earth to complete one revolution about its axis of rotation. This type of a satellite is always above the same spot on Earth.

5. How is time needed for one revolution about the orbit related to the speed of a satellite?

$$N = \frac{d}{t} = \frac{2\pi r}{T} \Rightarrow \text{as } T \uparrow \quad N \downarrow$$

∴ Speed and period are inversely proportional.

6. Calculate the magnitude of escape velocity of the Earth.

$$m = 5.98 \times 10^{24} \text{ kg}$$

$$v_{\text{esc}} = \sqrt{\frac{2GM_E}{r}}$$

$$= \sqrt{\frac{2(6.67 \times 10^{-11})(5.98 \times 10^{24})}{6.38 \times 10^6}} = 1.1 \times 10^4 \text{ m/s}$$

7. If you wanted to calculate escape speed from another planet, what variables would change in the formula?

- the mass of the planet and its radius

